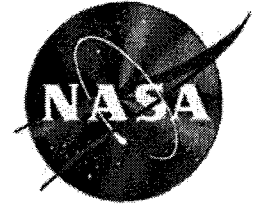


# MARS STRATIGRAPHY MISSION

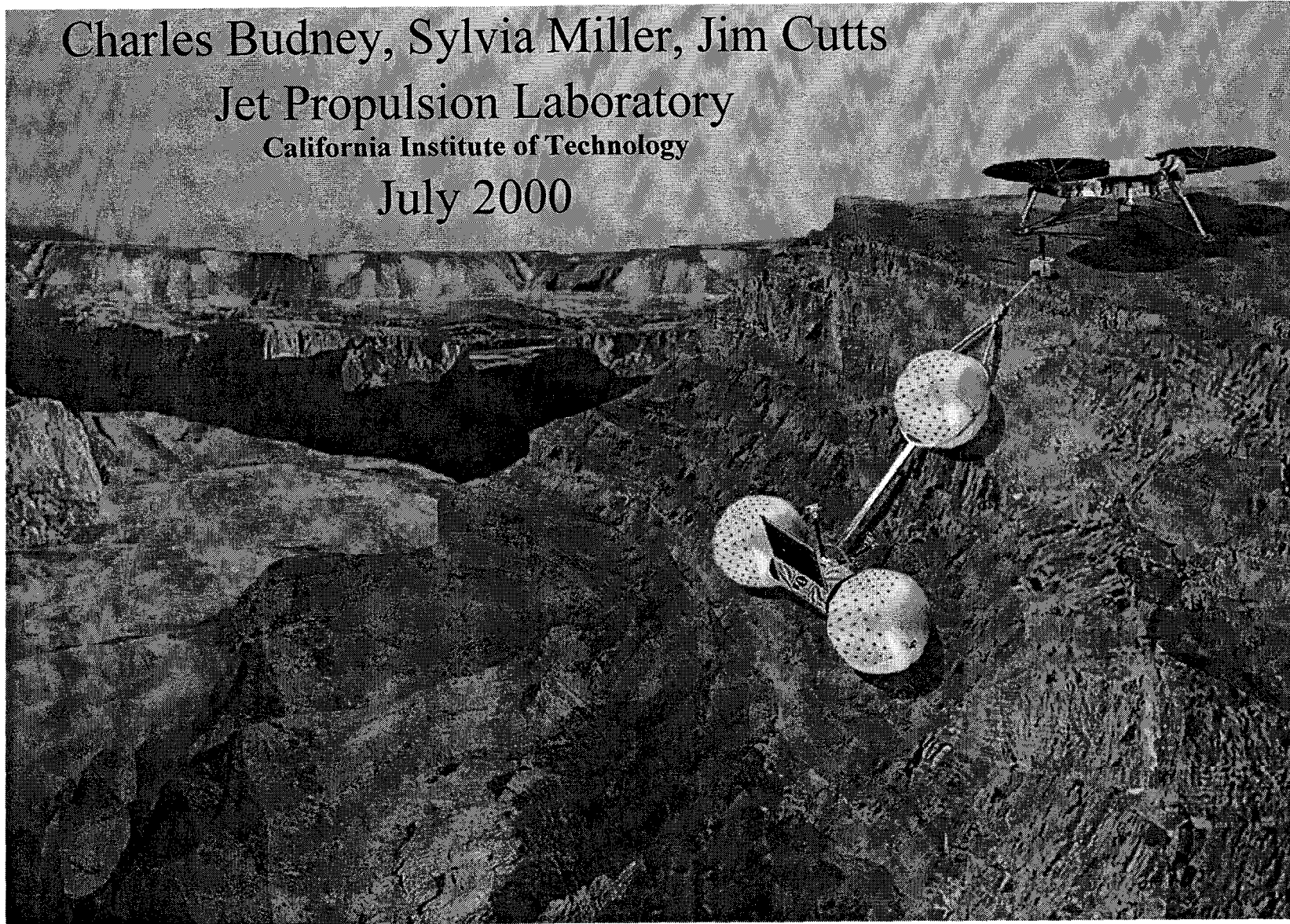
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Charles Budney, Sylvia Miller, Jim Cutts

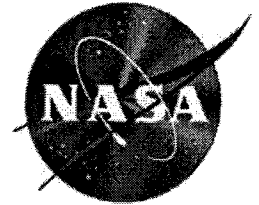
Jet Propulsion Laboratory  
California Institute of Technology

July 2000



# Mars Stratigraphy Mission

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## Science

### ► Science objectives

- Determine the geological history of the layered volcanic and sedimentary rocks of Valles Marineris
- Search for evidence of life within the deposits
- Elucidate the history of tectonic, volcanic, eolian and fluvial processes to characterize climate history
- Identify potential habitats for past and present Martian life

### ► Candidate Instruments

- Stereo multispectral imager
- Raman spectrometer
- X-Ray Florescence Spectrometer
- Age dating instrument
- Instrument arm
- Mini-corer
- Sample manipulation assembly
- Calibration targets for imager and x-ray spectrometer

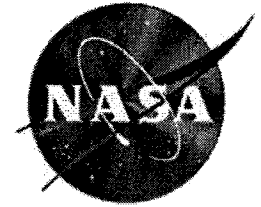
### ► Science operations

- Multispectral imaging, Raman, and XRF analyses every meter
- Sample collection for age dating every 100 meters



# Mars Stratigraphy Mission

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## Mission

### Scenario

- Landing site: 14S, 68W, near the southern canyon wall of Valles Marineris- 20 km circular landing zone
- Traverse to cliff top in <50 days
- Descend 2 km into the canyon on tether in 200 days
- Possible extended mission to canyon floor (6km, 400 days)

### Geometry

- 10 km landing error
- Arrive and operate while the sun is north of about  $-10^\circ$  declination

### Trajectory

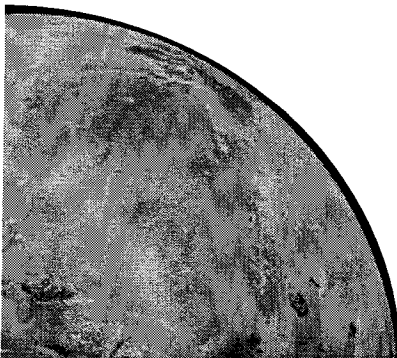
- Type IV
- $C_3$  of  $12 \text{ km}^2/\text{s}^2$
- Flight time 30 months
- Arrival  $V_\infty \sim 6 \text{ km/s}$

### Delta 7925

- LV capability  $\sim 975 \text{ kg}$  for this  $C_3$

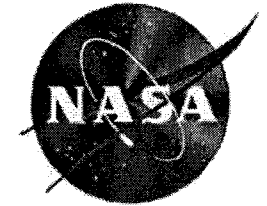
### Earliest Launch dates

- April 2007 (assumes 20 day launch period)



# Mars Stratigraphy Mission

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## Spacecraft

### ► Performance attributes

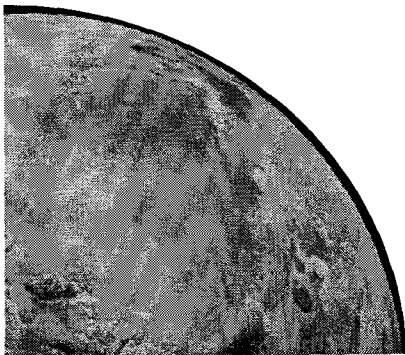
- Land within 10 km of cliff
- 20 km overland mobility
- 6 km of cable to lower rover down cliff
- Rover handles slopes from 0 - 90 degrees

### ► Flight system elements

- Inflatable rover
  - ♦ Solar powered
- Moderate lander
- Entry system
  - ♦ Direct entry

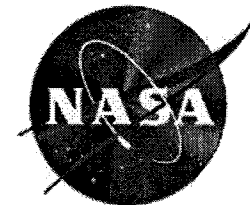
### ► Margins

- 30% mass/power contingency carried in design study
- 40% (400 kg) mass margin



# Mars Stratigraphy Mission

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## Technology / Infrastructure

### ► 2003 Technology Cutoff

### ► Critical Technology Needs

- Precision navigation and landing
- Long range mobility over hazardous terrain.
- In situ instrumentation
- Telecommunications
- Light weight drilling and rock sampling devices

### ► Candidate Technology Demos

- Precision navigation and landing (laser)
- Long range mobility over hazardous terrain.

### ► Infrastructure Needs

- Relay Orbiter(s)

### ► Acknowledgements

- Team X (study 8-99)
- Kerry Nock, JJ Wu, Dave Farless, Bob Balaram, Steve Townes

